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GRADIOMETER-AIDED RAPID GRAVITY SURVEY SYSTEM. (U)

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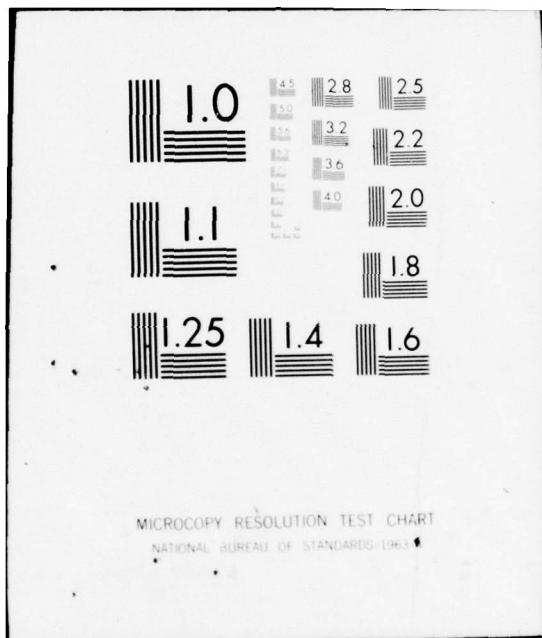
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ETL-0112

SP-957-1-1

GRADIOMETER-AIDED RAPID  
GRAVITY SURVEY SYSTEM

18 October 1976  
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Prepared for:

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) <i>This report considers a mobile vehicle equipped with both an inertial positioning system (IPS) and a gradiometer. For suitable gradiometer-aiding configurations, the following variables are determined: (1) Real-Time vs Post-Mission Data Processing, (2) Presence or absence of Terminal Calibration Data, (3) Continuous Time vs Halted Vehicle Gradiometer Operation, (4) Gradiometer Errors, (5) Zero Velocity and Gradiometer Calibration Stops, and (6) Gyro and accelerometer Errors. This report concluded that one sec or better gradiometer-aided</i>		
(continued) →		

4PZ

20. continued

RGSS performance in open traverse is unlikely without vertical deflection end calibration. In addition, the keynote of successful RGSS/Gradiometer integration will be control and compensation of system bias and low frequency error sources.

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## FOREWORD

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# OVERVIEW

R-25236

- RAPID GRAVITY SURVEY SYSTEM SCENARIO AND PERSPECTIVE ON GRADIOMETRY
- VERTICAL DEFLECTIONS FROM GRADIOMETER DATA ALONE
- GRADIOMETER-AIDED RGSS SIMULATION DESCRIPTION
- COVARIANCE SIMULATION RESULTS (FIRST PHASE)
- PRELIMINARY CONCLUSIONS AND A LOOK AHEAD



RAPID GRAVITY SURVEY SYSTEM (RGSS)  
SCENARIO AND PERSPECTIVE ON GRADIOMETRY

R-25237

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## PROBLEM STATEMENT

R-25238

CONSIDER A MOBILE VEHICLE EQUIPPED WITH BOTH AN INERTIAL POSITIONING SYSTEM (IPS)\* AND A GRADIOMETER. FOR SUITABLE GRADIOMETER-AIDING CONFIGURATIONS, DETERMINE THE EFFECT OF THE FOLLOWING VARIABLES:

- REAL-TIME vs POST-MISSION DATA PROCESSING
- PRESENCE OR ABSENCE OF TERMINAL CALIBRATION DATA
- CONTINUOUS TIME vs HALTED VEHICLE GRADIOMETER OPERATION
- GRADIOMETER ERRORS
- ZERO VELOCITY AND GRADIOMETER CALIBRATION STOPS
- GYRO AND ACCELEROMETER ERRORS

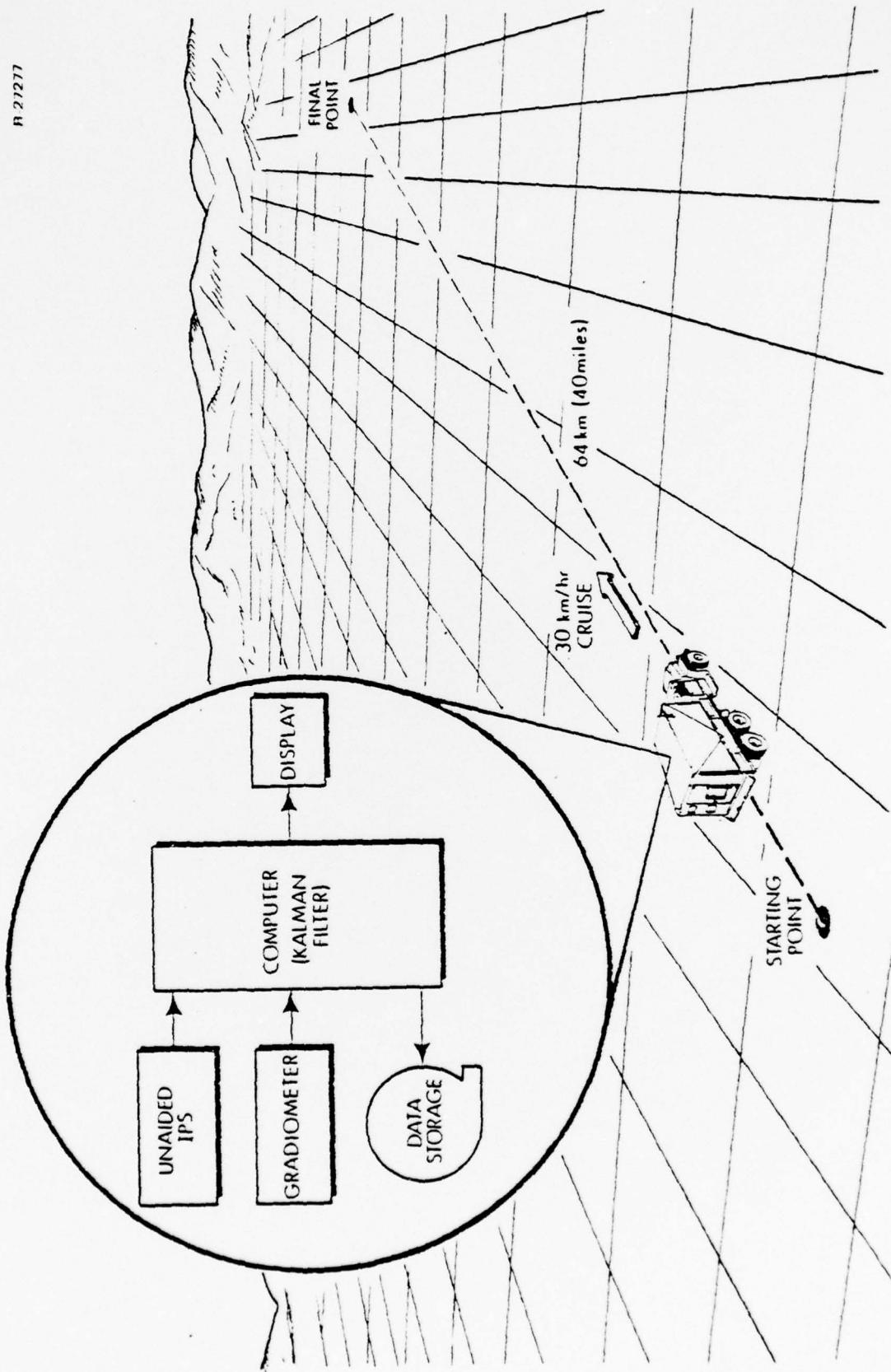
\*CURRENTLY CONFIGURED AS A LITTON LN15 INERTIAL NAVIGATION SYSTEM WITH IMPROVED (A-1000) VERTICAL CHANNEL ACCELEROMETER

REFERENCE: Huddle, J.R., "Navigation to Surveying Accuracy With an Inertial System," Bicentennial National Aerospace Symposium, Warminster, PA, April 1976.

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# TYPICAL SURVEY CONFIGURATION

R 21277



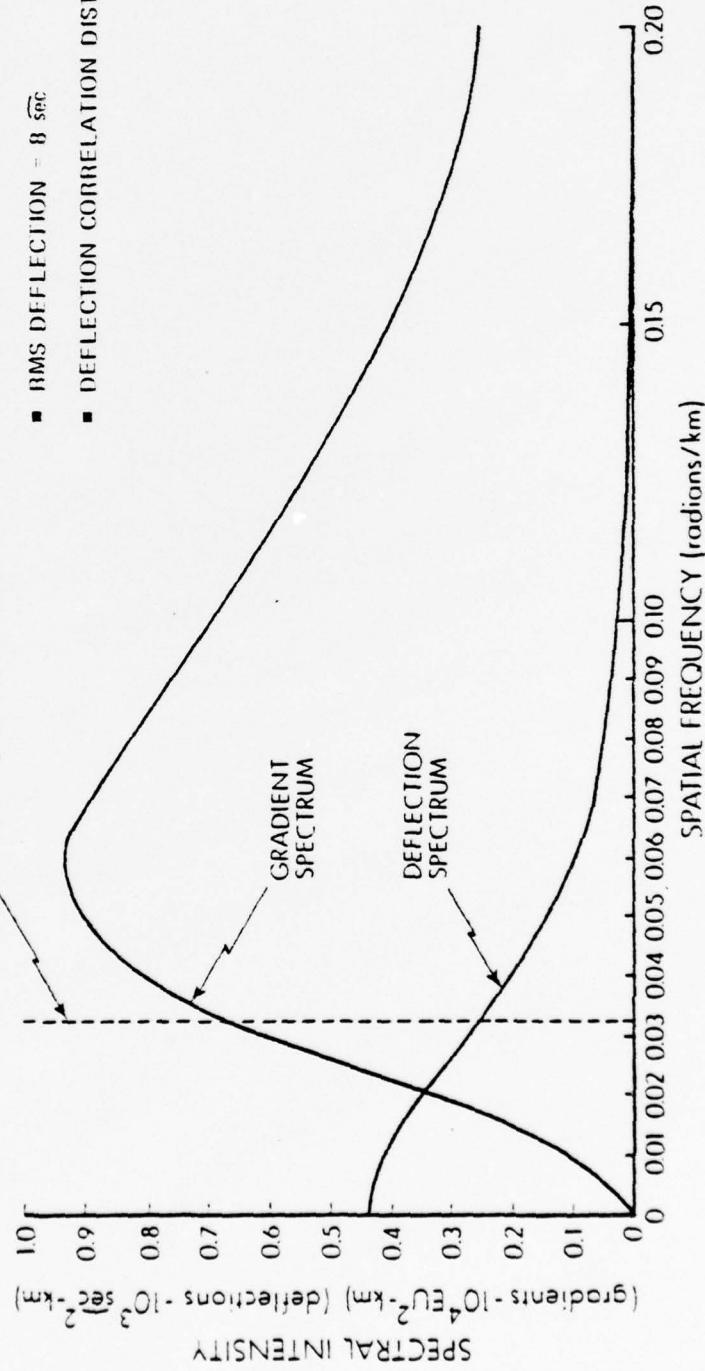
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# GRAVITY QUANTITY POWER SPECTRA

R 25274

- SECOND ORDER MARKOV CROSS-TRACK DEFLECTION MODEL
- GRADIENT AUTO SPECTRUM IS FOR ALONG TRACK, CROSS TRACK TENSOR ELEMENT
- RMS DEFLECTION =  $8 \text{ sec}^{\frac{1}{2}}$
- DEFLECTION CORRELATION DISTANCE = 37 km

LONG-WAVELENGTH "LIMIT"  
FOR 64 km (40 mile) TRAVERSE



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## MULTISENSOR VIEW OF GRADIOMETER-AIDED RGSS

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- GRADIOMETER MEASURES SHORT WAVELENGTH FEATURES OF THE GRAVITY FIELD
- IPS MEASURES LONG WAVELENGTH FEATURES
- OPTIMAL COMBINATION OF BOTH SETS OF MEASUREMENTS (KALMAN FILTERING OR SMOOTHING) YIELDS BEST (IN A LEAST SQUARES SENSE) DEFLECTION RECOVERY

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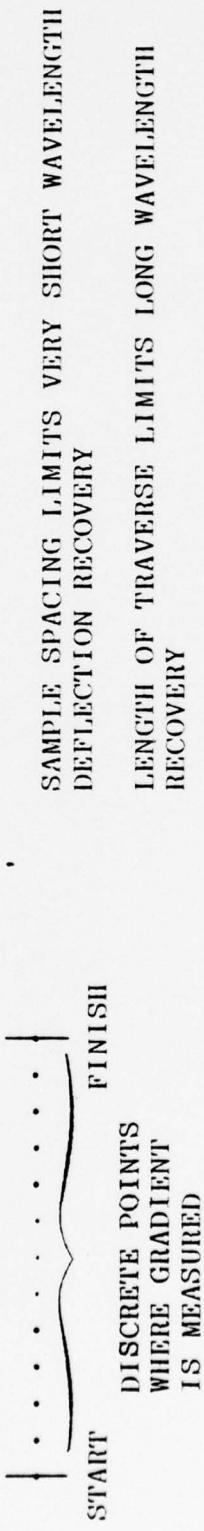
**VERTICAL DEFLECTIONS FROM  
GRADIOMETER DATA ALONE**

R-25240

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# NON-MOBILE, GRADIOMETER-ALONE VERTICAL DEFLECTION SURVEY

R-25241



- SIMPLEST IMAGINABLE GRADIOMETER SURVEY APPROACH
- DOES NOT TAKE ADVANTAGE OF LONG WAVELENGTH INFORMATION AVAILABLE FROM INERTIAL SYSTEM
- DEFLECTION RECOVERY RESULTS NOT LIMITED BY HIGH FREQUENCY GRADIOMETER ACCURACY

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**VERTICAL DEFLECTION RECOVERY WITH DISCRETE, NON-MOBILE  
GRADIOMETER MEASUREMENTS ONLY**

R-25242

NUMBER OF DISCRETE GRADIOMETER MEASUREMENTS (Equally Spaced)	GRADIOMETER NOISE		RMS DEFLECTION ESTIMATION ERROR FOR ENTIRE TRAVERSE (sec)
	WHITE NOISE (EU)*	BIAS (EU)	
10 (7.1 km spacing)	0	0	4.9
10 (7.1 km spacing)	1.0	0	5.4
10 (7.1 km spacing)	0	1.0	6.9
41 (1.6 km spacing)	1.0	0	4.3
41 (1.6 km spacing)	0	0	3.8

- RMS VERTICAL DEFLECTION =  $8 \sqrt{\text{sec}}$
- DEFLECTION CORRELATION DISTANCE = 37 km
- TRAVERSE DISTANCE = 64 km (40 mi)
- VEHICLE "DWELL TIME" AT EACH MEASUREMENT SITE = 100 seconds
- OPTIMAL, POST-MISSION DATA PROCESSING (SMOOTHING)

CONCLUSION: GRADIOMETER MUST OPERATE IN MULTISENSOR CONTEXT

\*TEN SECOND MOVING WINDOW AVERAGE

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# GRADIOMETER-AIDED RGSS SIMULATION DESCRIPTION

R-25243

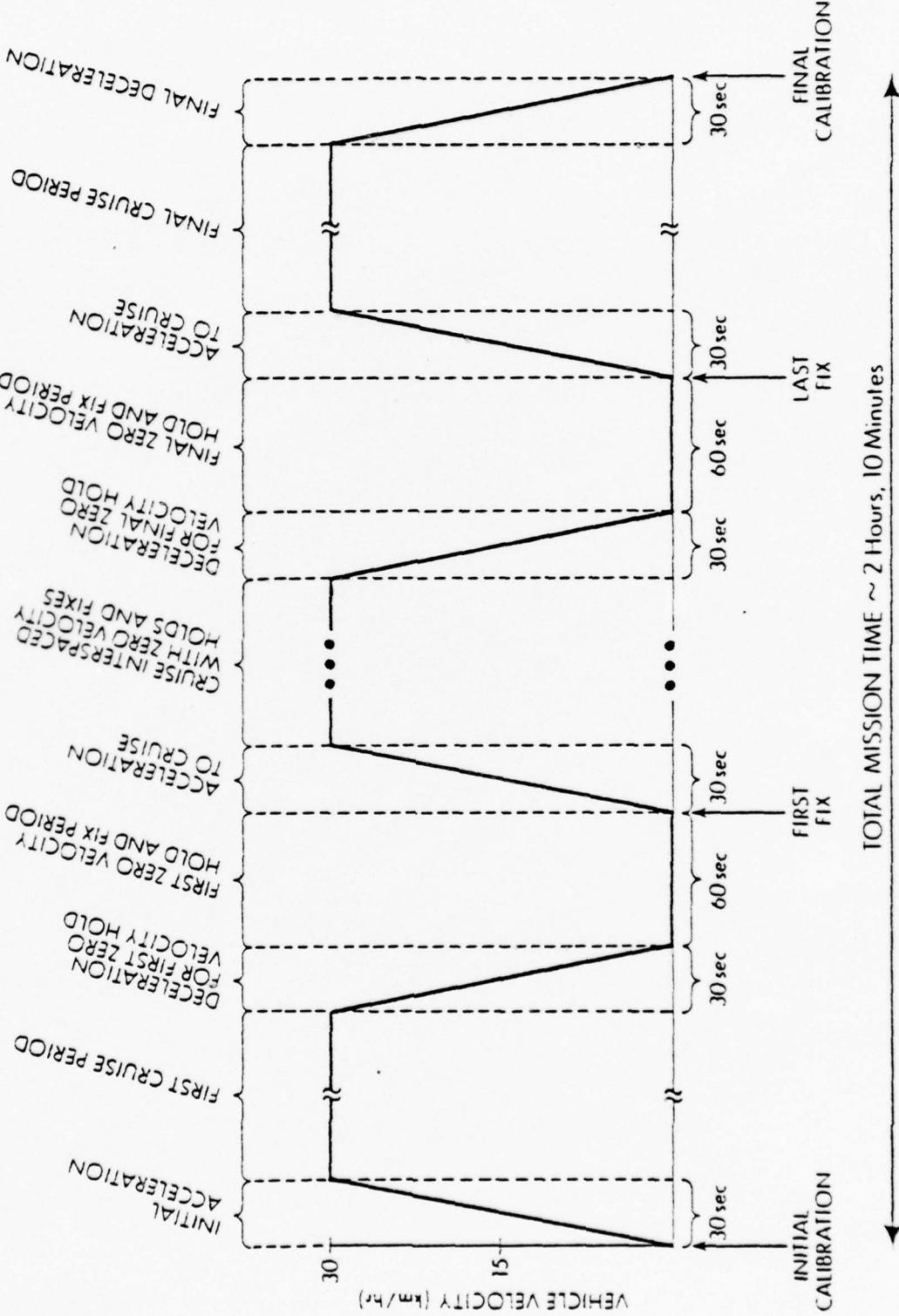
## SIMULATION FEATURES

- R-25244
- OPTIMAL KALMAN PROCESSING OF DATA
  - ERROR COVARIANCE HISTORY OF RESIDUAL GRAVITY, VELOCITY AND POSITION ERRORS ALONG SURVEY TRACK
  - SINGLE-CHANNEL, "QUICK-LOOK" ANALYSIS
  - PROVISION FOR INCLUSION OF GYRO ERRORS
  - PROVISION FOR ZERO VELOCITY UPDATES (VELOCITY AND GRADIOMETER "FIXES")
  - GRADIOMETER OUTPUT SUPPRESSED DURING ACCELERATION AND DECELERATION

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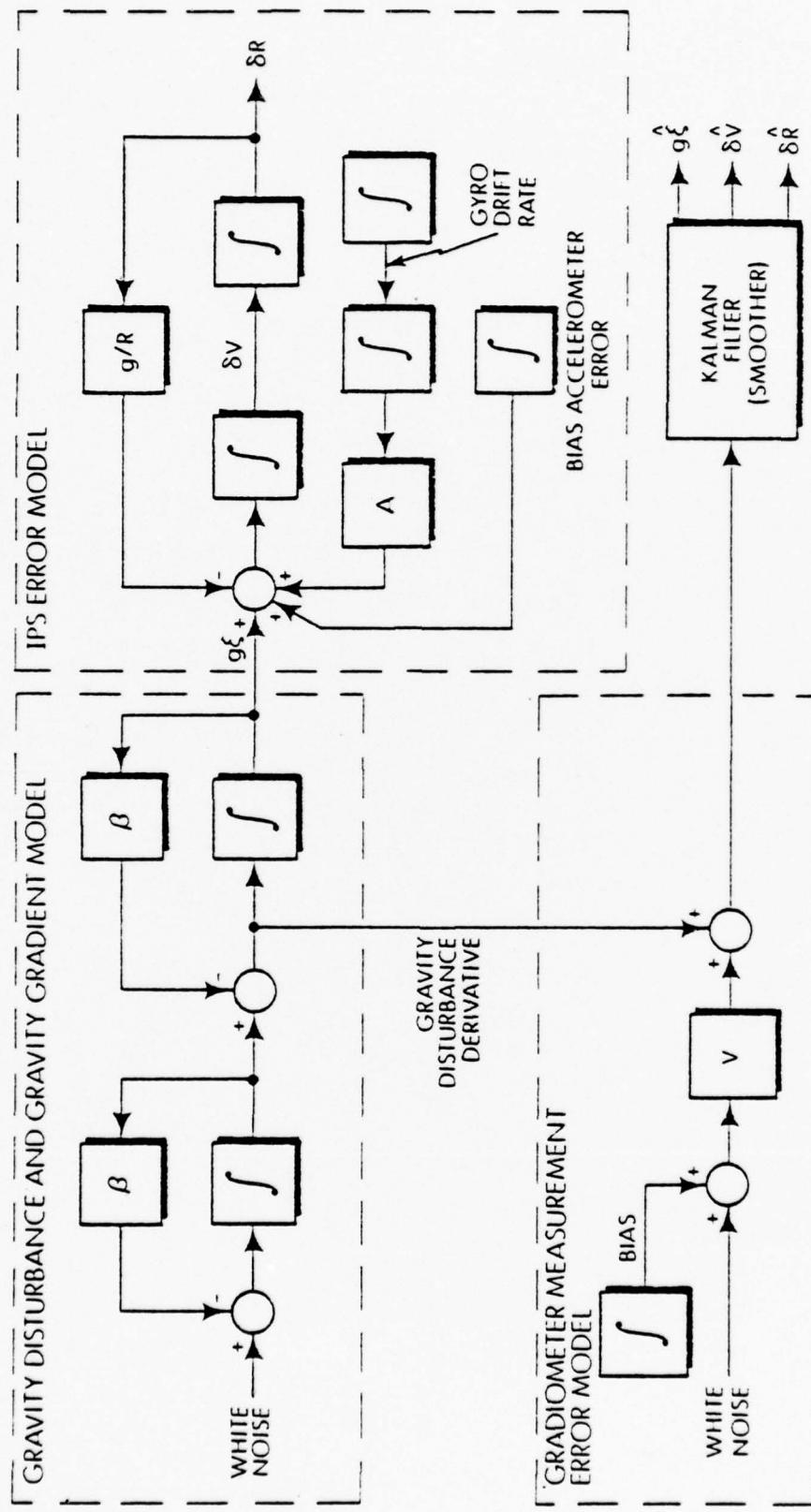
## MISSION PROFILE

R 27278



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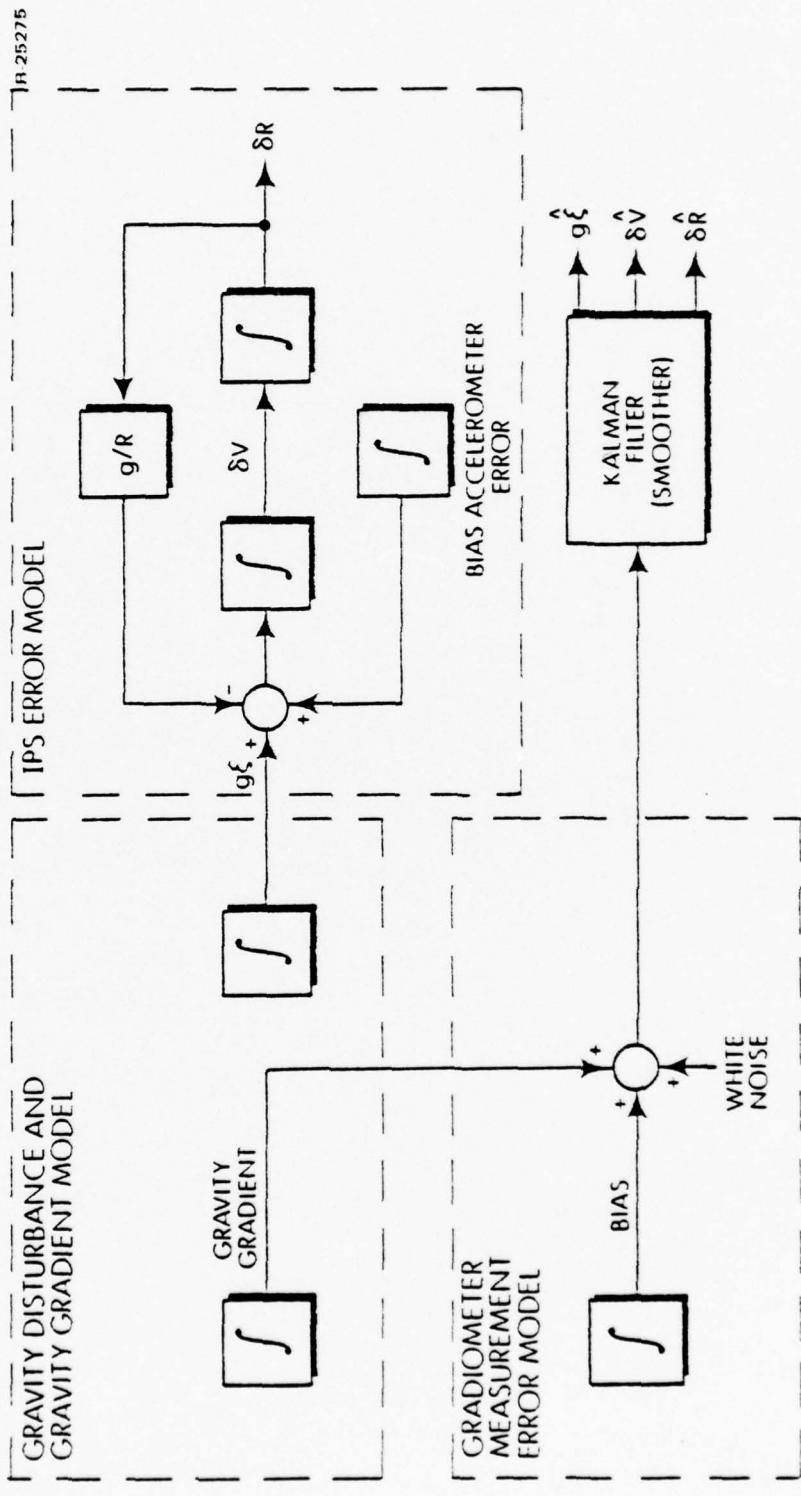
# COVARIANCE SIMULATION MODEL FOR MOVING VEHICLE WITH MOBILE GRADIOMETER



$\beta$  = DEFLECTION CORRELATION FREQUENCY  
 $\Lambda$  = GRAVITY  
 $\delta v$  = EARTH RADIUS  
 $\delta r$  = ESTIMATED QUANTITY  
 $\xi$  = CROSS-TRACK VERTICAL DEFLECTION

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# SIMULATION MODEL FOR STATIONARY VEHICLE AND GRADIOMETER



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# SUMMARY OF SIMULATION CONSTANTS

## SURVEY

TRAVERSE DISTANCE = 64 km (40 miles)  
VEHICLE CRUISE SPEED = 30 km/hr  
ACCELERATION PERIODS = 30 seconds each

## BACKGROUND GRAVITY DISTURBANCE FIELD (MARKOV)

VERTICAL DEFLECTION = 8 sec rms  
DEFLECTION CORRELATION DISTANCE = 37 km

## SYSTEM FIELD CALIBRATION AND "FIXES" (When Applicable)

TERMINAL POINT DWELL TIME = 1000 seconds  
ZERO VELOCITY UPDATE DWELL TIME = 60 seconds

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# SUMMARY OF SIMULATION VARIABLES (FIRST PHASE SIMULATION RESULTS)

R-25246

## GRADIOMETER ERRORS

WHITE NOISE = 10 EU\*  
BIAS = 0, 10 EU

## IPS ERRORS

RESIDUAL CROSS-TRACK ACCELEROMETER ERROR = 4 sec  
VERTICAL COMPONENT OF POLAR GYRO DRIFT RATE = 0

## CALIBRATION AND FIX TAKING

CALIBRATION QUANTITIES AT END POINTS - POSITION, VELOCITY,  
DEFLECTION, GRAVITY GRADIENT  
NUMBER OF STOPS (VELOCITY AND GRADIENT FIXES) - 0

\*TEN SECOND MOVING WINDOW AVERAGE

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COVARIANCE SIMULATION RESULTS  
(FIRST PHASE)

R-25247

# FIRST PHASE SIMULATION STUDY CASES

R-25248

## RGSS DEFLECTION RECOVERY AND NAVIGATION ACCURACY WITHOUT GRADIOMETER-AIDING (REAL-TIME AND POST-MISSION)

CASE	QUANTITIES INCLUDED IN HIGH QUALITY ENDPOINT CALIBRATION
1	$\delta R$ , $\delta V$
2	$\delta R$ , $\delta V$ , $\xi$
3	$\delta R$ , $\delta V$ , $\xi$ , $\partial \xi / \partial x^*$

## RGSS DEFLECTION RECOVERY AND NAVIGATION ACCURACY WITH MOVING-BASE GRADIOMETER-AIDING (REAL-TIME AND POST-MISSION)

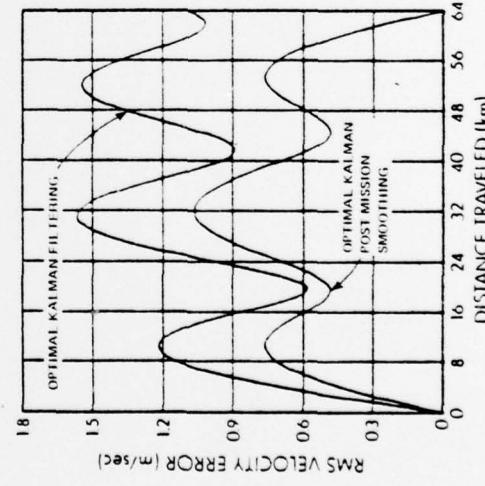
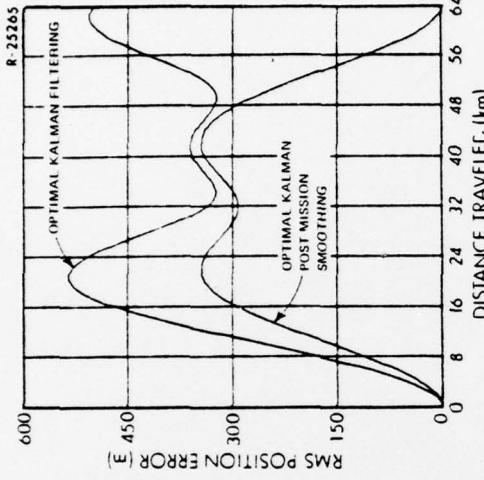
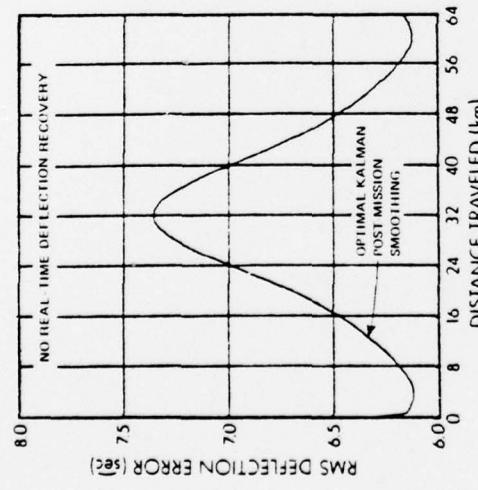
CASE	QUANTITIES INCLUDED IN HIGH QUALITY ENDPOINT CALIBRATION	GRADIOMETER ENDPOINT CALIBRATION† ASSUMED
4	$\delta R$ , $\delta V$ , $\xi$	NO
5	$\delta R$ , $\delta V$	YES
6 (No bias errors)	$\delta R$ , $\delta V$	NO

\*  $\partial \xi / \partial x$  = GRAVITY DISTURBANCE GRADIENT

† GRADIOMETER OUTPUT IS AVERAGED DURING 1000 SECOND CALIBRATION INTERVAL

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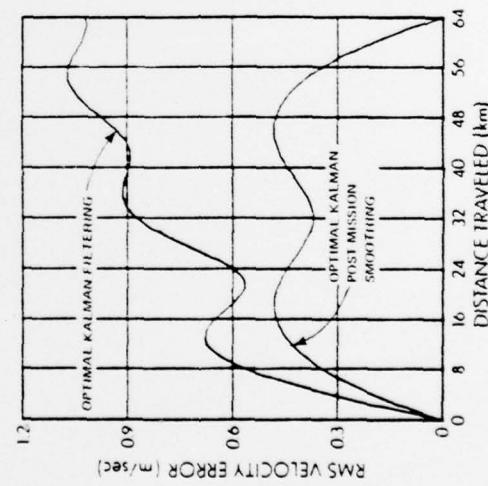
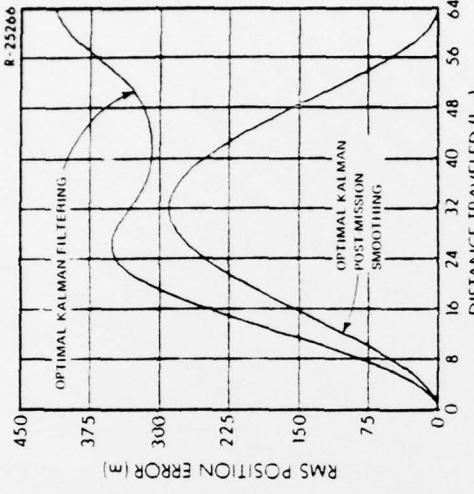
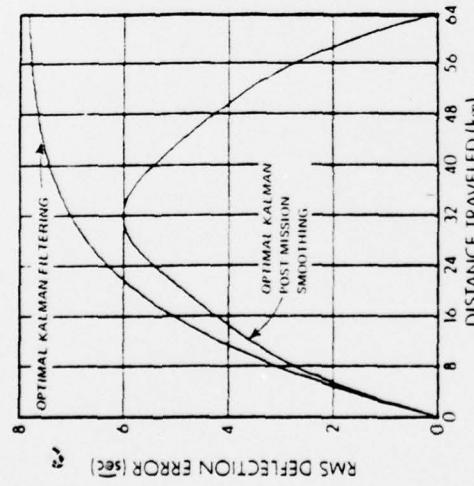
# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF ENDPOINT CALIBRATION DATA



CASE 1

- NO GRADIOMETER AIDING
- PRECISION ENDPOINT CALIBRATION OF
  - POSITION
  - VELOCITY

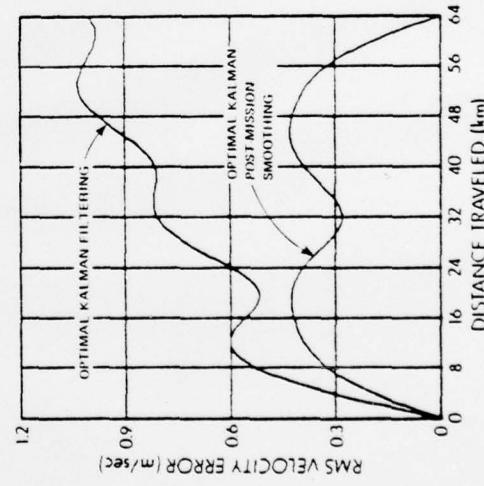
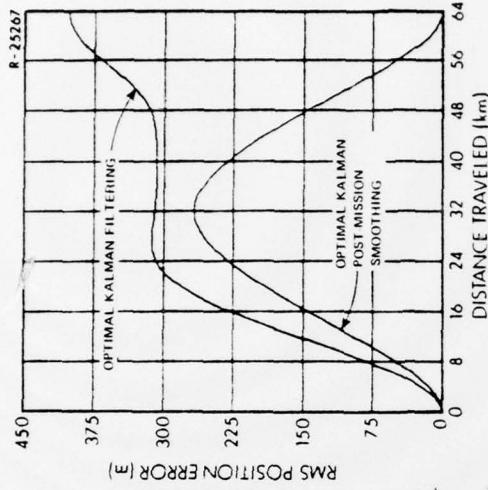
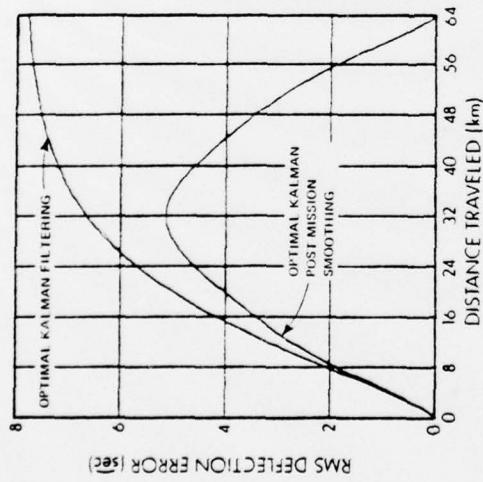
# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF ENDPOINT CALIBRATION DATA



**CASE 2**

- NO GRADIOMETER AIDING
- PRECISION ENDPOINT CALIBRATION OF
  - POSITION
  - VELOCITY
  - VERTICAL DEFLECTION

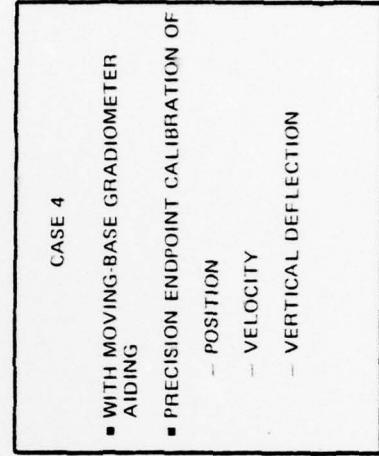
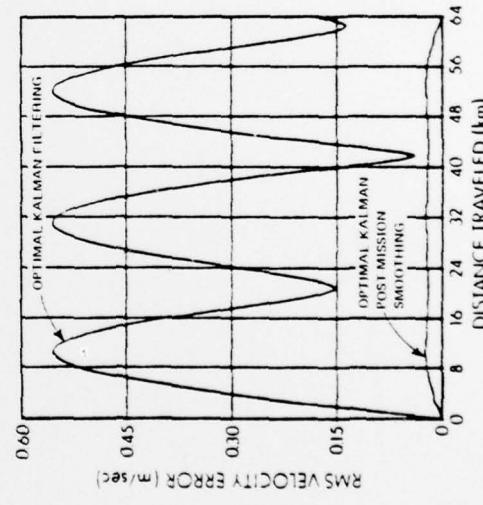
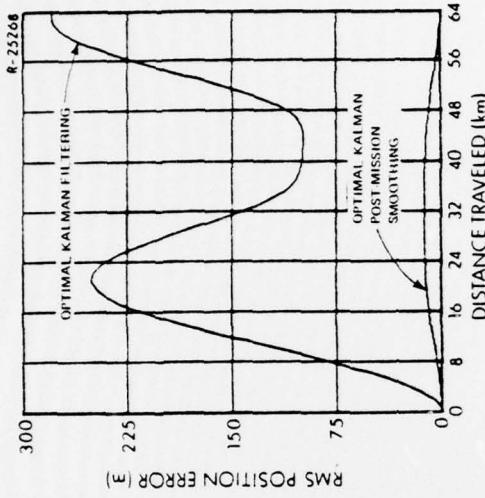
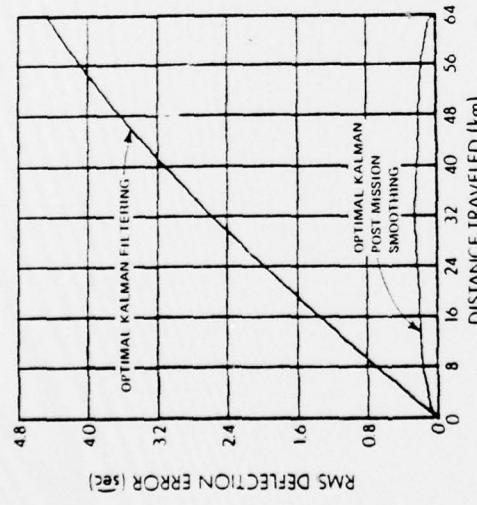
# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF ENDPOINT CALIBRATION DATA



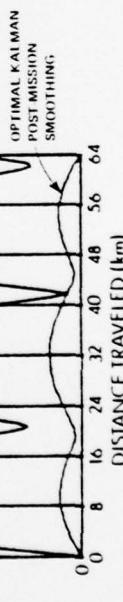
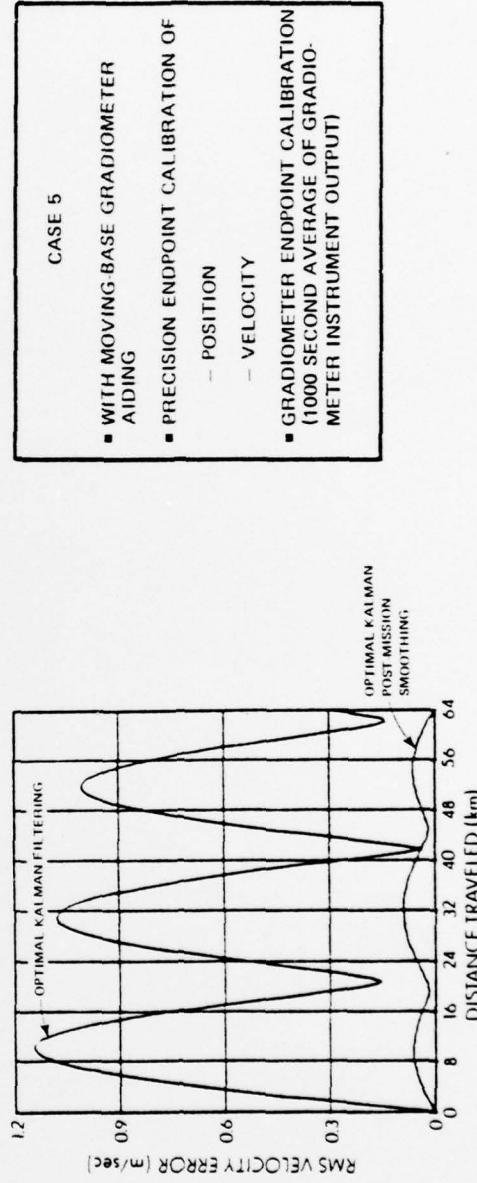
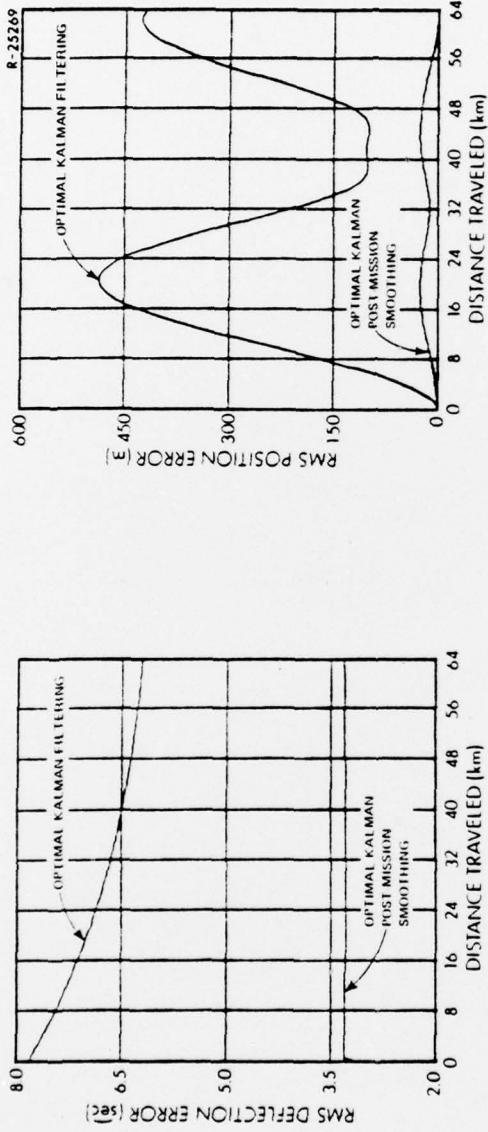
CASE 3

- NO GRADIOMETER AIDING
- PRECISION ENDPOINT CALIBRATION OF
  - POSITION
  - VELOCITY
  - VERTICAL DEFLECTION
  - GRAVITY GRADIENT

# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF GRADIOMETER AND ENDPOINT CALIBRATION DATA

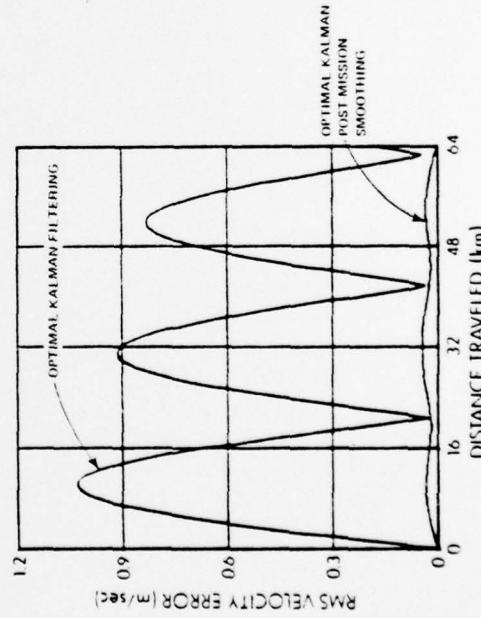
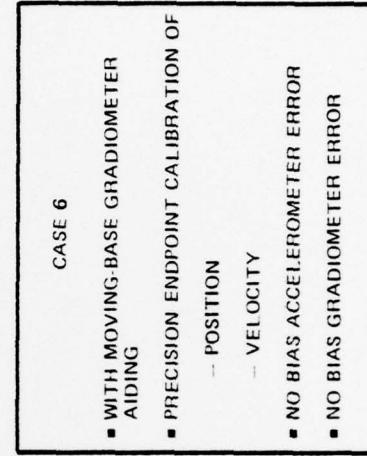
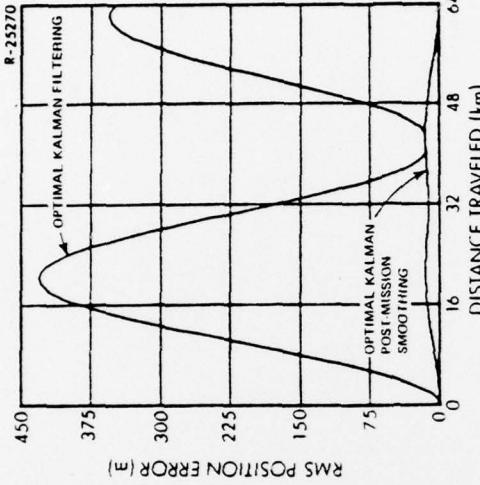
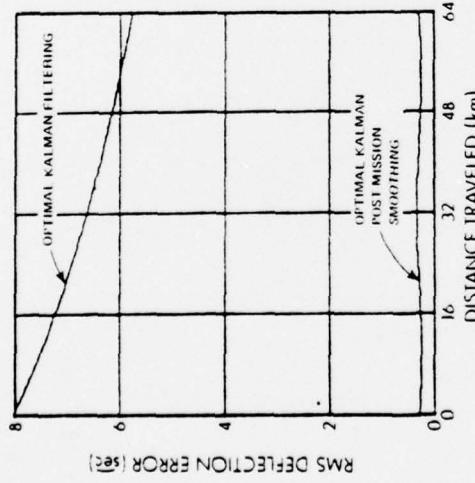


# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF GRADIOMETER AND ENDPOINT CALIBRATION DATA



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# RGSS SURVEY ERRORS FOR OPTIMAL KALMAN PROCESSING OF GRADIOMETER AND ENDPOINT CALIBRATION DATA



# SUMMARY OF SIMULATION RESULTS

CASE	TYPE OF END POINT CALIBRATION	GRADIOMETER ERRORS	RMS ERRORS OVER ENTIRE SURVEY			OPTIMAL SMOOTHING		
			REAL-TIME DEFLECTION (sec)	OPTIMAL POSITION (m)	VELOCITY (m/sec)	DEFLECTION (sec)	POSITION (m)	VELOCITY (m/sec)
1	$\delta R, \delta V$	NO GRADIOMETER	8.0	381	1.1	6.6	252	0.7
2	$\delta R, \delta V, \xi$	NO GRADIOMETER	6.4	299	0.8	4.3	181	0.4
3	$\delta R, \delta V, \xi, \frac{\partial \xi}{\partial x}$	NO GRADIOMETER	6.1	287	0.8	3.5	167	0.3
4	$\delta R, \delta V, \xi$	$\left\{ \begin{array}{l} 10 \text{ EU} \\ \text{WHITE} \\ \text{NOISE*} \end{array} \right\}$ AND $\left\{ \begin{array}{l} 10 \text{ EU} \\ \text{BIAS} \end{array} \right\}$	2.8	174	0.4	0.2	8.3	0.02
5	$\delta R, \delta V, \frac{\partial \xi}{\partial x}$ FROM GRADIOMETER	GRADIOMETER BIAS	6.8	295	0.8	3.3	17.5	0.05
6 <sup>†</sup>	$\delta R, \delta V$	10 EU WHITE NOISE* ONLY	6.8	254	0.6	0.3	10.7	0.03

\* 10 SECOND MOVING WINDOW AVERAGE

<sup>†</sup> NOMINAL, 4 sec BIAS ACCELEROMETER ERROR SET TO ZERO

## OBSERVATIONS

R 25250

- PERFORMANCE OF MOBILE GRADIOMETER-AIDED IGSS VERY SENSITIVE TO ACCURACY OF TERMINAL DEFLECTION CALIBRATION
- VELOCITY AND GRAVITY GRADIENT FIXES ASSOCIATED WITH PERIODIC ZERO VELOCITY HOLDS EXPECTED TO PROVIDE SIGNIFICANT REDUCTION IN POSITION AND VELOCITY ERRORS; SMALL REDUCTION IN DEFLECTION ERRORS
- SIMILAR LOW FREQUENCY DYNAMICS OF DEFLECTIONS AND LOW FREQUENCY SYSTEM ERRORS CAUSE STRONG SENSITIVITY OF ESTIMATION ERROR TO BIAS-LIKE ERROR SOURCES

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PRELIMINARY CONCLUSIONS  
AND A LOOK AHEAD

R-25249

## PRELIMINARY CONCLUSIONS

R-25262

ONE SEC OR BETTER GRADIOMETER-AIDED RGSS PERFORMANCE  
IN OPEN TRAVERSE IS UNLIKELY WITHOUT VERTICAL,  
DEFLECTION END CALIBRATION

KEYNOTE OF SUCCESSFUL RGSS/GRADIOMETER INTEGRATION WILL  
BE CONTROL AND COMPENSATION OF SYSTEM BIAS AND LOW  
FREQUENCY ERROR SOURCES

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## A LOOK AHEAD

R-25251

CURRENT AND SOON-TO-BE-DEVELOPED SIMULATION SOFTWARE WILL PROVIDE A FIRST-LOOK UNDERSTANDING OF THE FOLLOWING EFFECTS

- ULTIMATE PERFORMANCE POSSIBLE WITH RGSS OPTIMALLY AIDED BY NON-MOBILE GRADIOMETER
- RGSS PERFORMANCE IMPROVEMENT WITH ZERO-VELOCITY HOLDS (FIXES) IN ADDITION TO MOVING-BASE GRADIOMETER-AIDING
- DEGRADATION DUE TO INSTRUMENT ERRORS
  - DIFFERENT GRADIOMETER NOISE LEVELS
  - GYRO DRIFT
- INSIGHT INTO DEGRADATION DUE TO IPS ERRORS NOT OBSERVABLE AT ZERO-VELOCITY (e.g., ACCELEROMETER SCALE FACTOR)

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